



# **An Investigation into the Root Causes of O-Ring Adhesion**

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Group 75**

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WPI Advisor**

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# Outline

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**Introduction**

**Test Results**

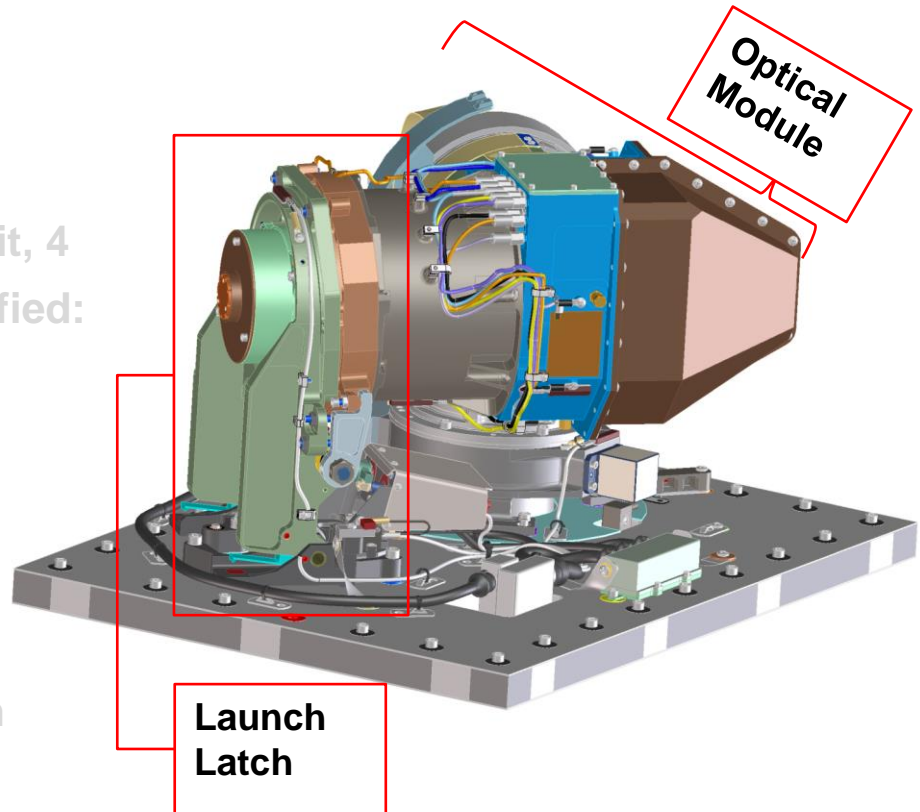
**Part Selection**

**Recommendations**



# The Space Terminal

- **The space terminal is a space-based high data rate laser communication system**
- **Launch Latch and interface**
  - Restrains telescope during launch
  - Contamination seal during ascent
- **To ensure the Launch Latch opens in orbit, 4 Launch Latch components must be specified:**
  - O-ring
  - Mating Interface
  - Lower Torsion Springs
  - Upper Vlier Spring Plunger
- **Launch Latch design criteria**
  - The load on the HOPAs and Pawl Arm do not yield or break during launch
  - Launch Latch opens once on orbit

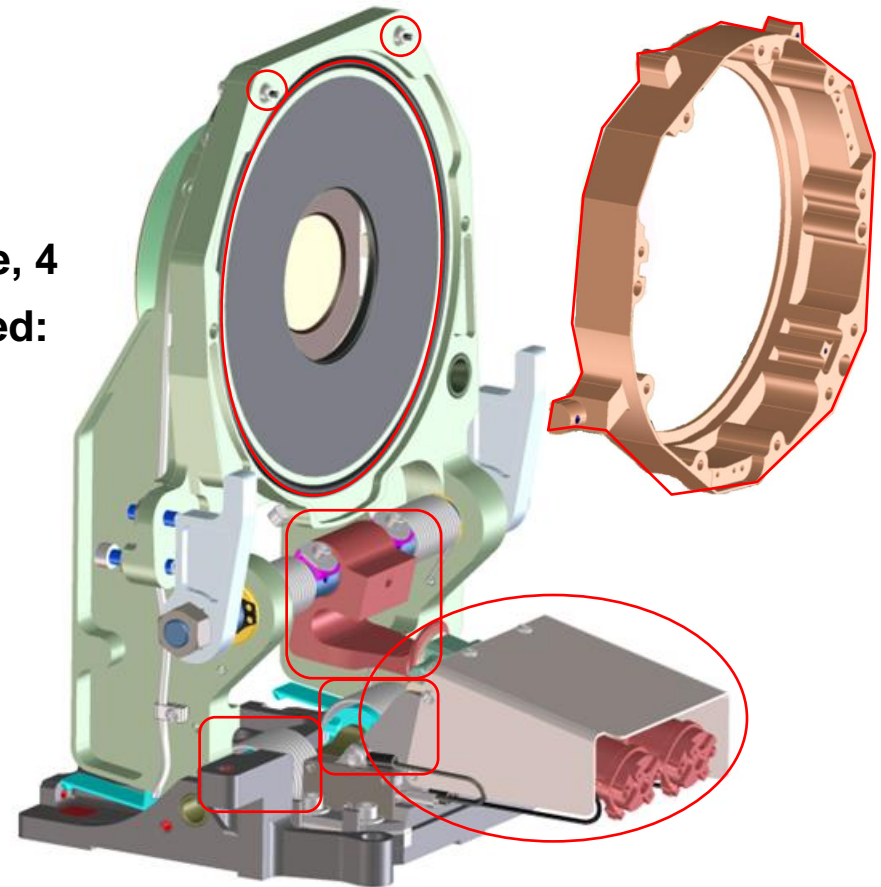


**Select components for Launch Latch design to ensure survival of hardware and successful opening**



# The Space Terminal

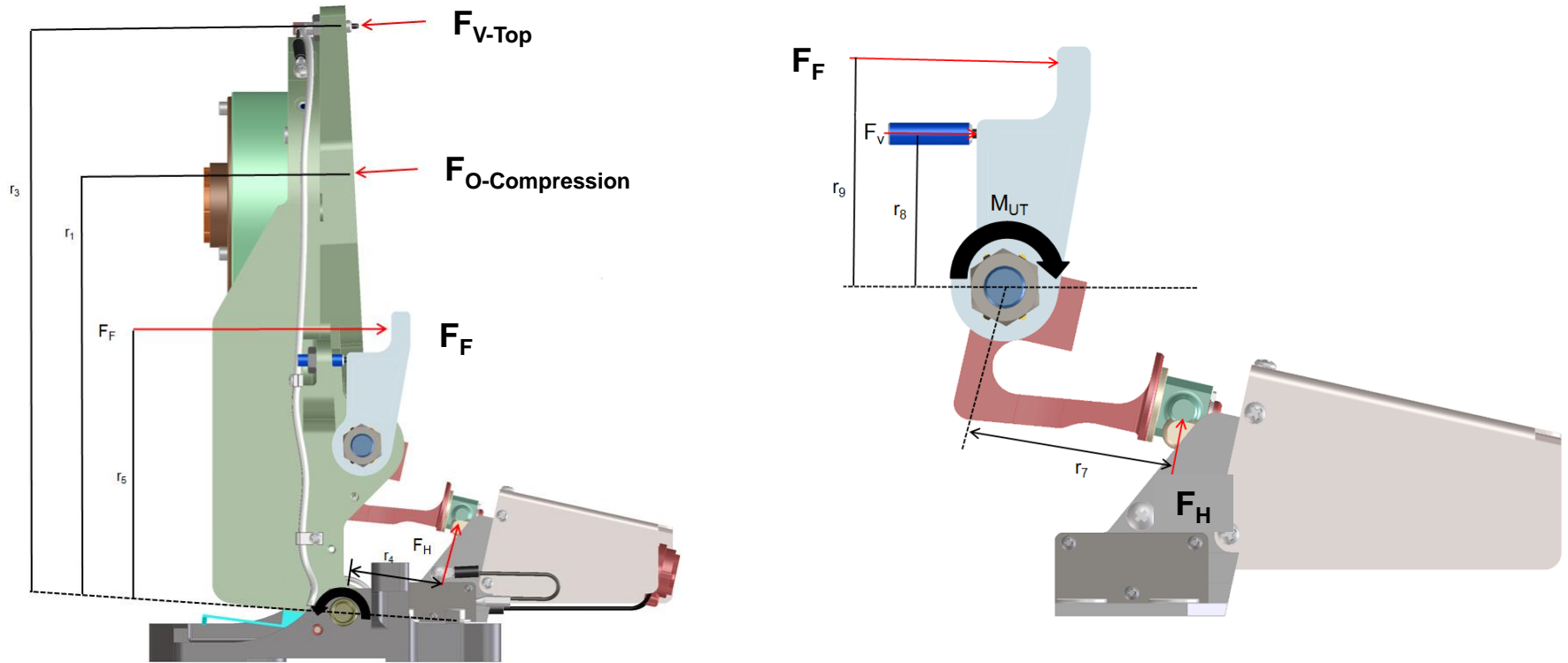
- The space terminal is a space-based high data rate laser communication system
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**Select components for Launch Latch design to ensure survival of hardware and successful opening**



# Problem: Ensure Latch Parts Survive



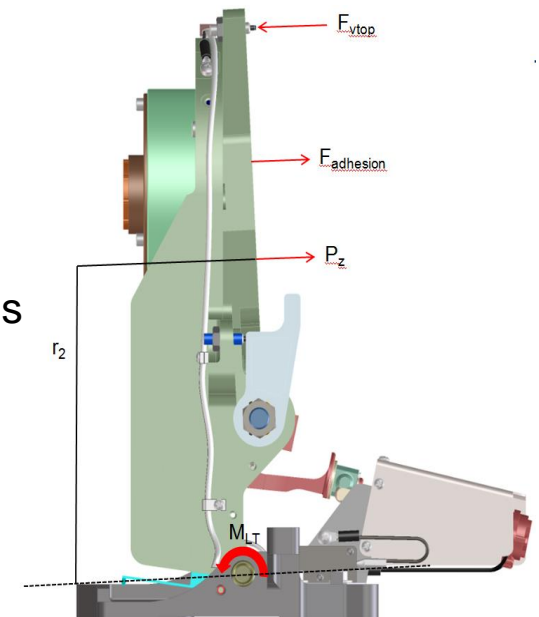
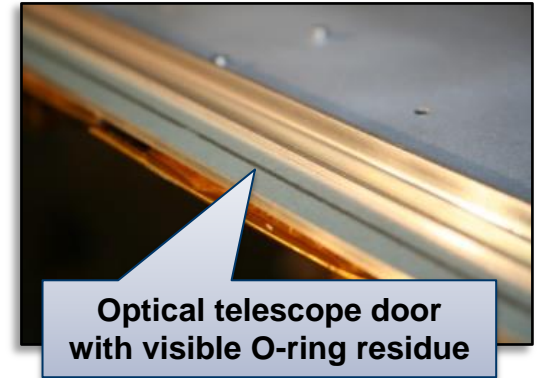
- Sum of the moments to find relations between  $F_F$  and  $F_H$
- Selection of O-Ring, Vliers, and Lower Torsion Springs determine  $F_H$
- $F_{Load} < F_{Vibration} + F_{Preload} + F_{HOPA}$

Select components for Launch Latch design to ensure survival of hardware during launch



# Problem: Ensure Latch Opens

- **What can prevent opening?**
  - Parasitic latch pin force  $P_z$  (given)
    - Well understood and taken into account
  - O-ring adhesion
    - Fundamental mechanics not well understood
    - Time must be spent on this problem
- **Method to ensuring successful opening:**
  - Test O-ring adhesion
    - Quantify promising candidate O-rings and interfaces in search for a zero adhesion pairing
  - Select parts to overcome adhesion force
    - 1.25 safety factor applied to maximum adhesion force from testing



**Testing of adhesion force and proper part selection will ensure successful latch opening**



# O-Rings and Interfaces

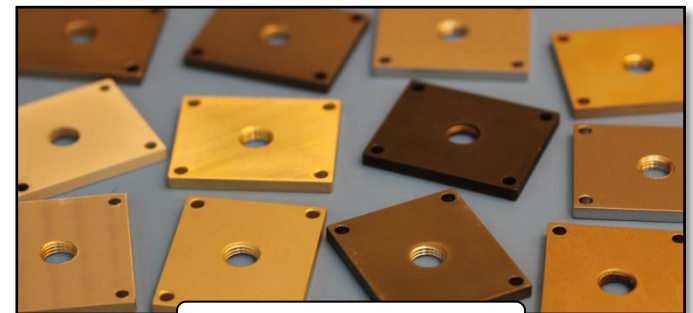
O-Ring	Type	Manufacturer
V0986	Fluorocarbon	Parker
S0899	Silicone Rubber	Parker
C0267	Polychloroprene	Parker
E1100	Ethylene Propylene Rubber	Parker
LM151	Fluorosilicone	Parker
S0469	Silicone Rubber	Parker
S0820	Silicone Rubber	Parker
CV2289*	RTV	MIT LL
RTV566*	RTV	MIT LL
SCV2585*	RTV	MIT LL
JaBar	Silicone Rubber	Ja-Bar
S7440	Silicone 0.05" Inner Diameter (ID)	Parker
S7440 FEP*	Silicone 0.05" ID FEP Encapsulated	MIT LL
Creavey 030	Silicone 0.03" ID FEP Encapsulated	Creavey
Creavey 050	Silicone 0.05" ID FEP Encapsulated	Creavey

\* Custom

Interface	Type	Manufacturer
Chem Film (CFM)	AL Finish	n/a
Chem Film, Polished (CFP)	AL Finish	n/a
Chem Film, Bead Blasted (CFB)	AL Finish	n/a
Silver Teflon Tape (TAPE)	AL Coating	n/a
Nedox NH1 (NH1)	AL Coating	General Magnaplate
Sanford Hardlube (SANF)	AL Coating	General Magnaplate
Dicronite (DICR)	AL Coating	Dicronite
Tufram HTR (HTR)	AL Coating	General Magnaplate
Invar, none (INV)	Invar Finish	n/a
Invar, Bead Blasted (INB)	Invar Finish	n/a
Nedox 615 (N615)	Invar Coating	General Magnaplate
Nedox SF-2 (SF2)	Invar Coating	General Magnaplate



**A few O-Rings We Tested**



**Interfaces Tested**

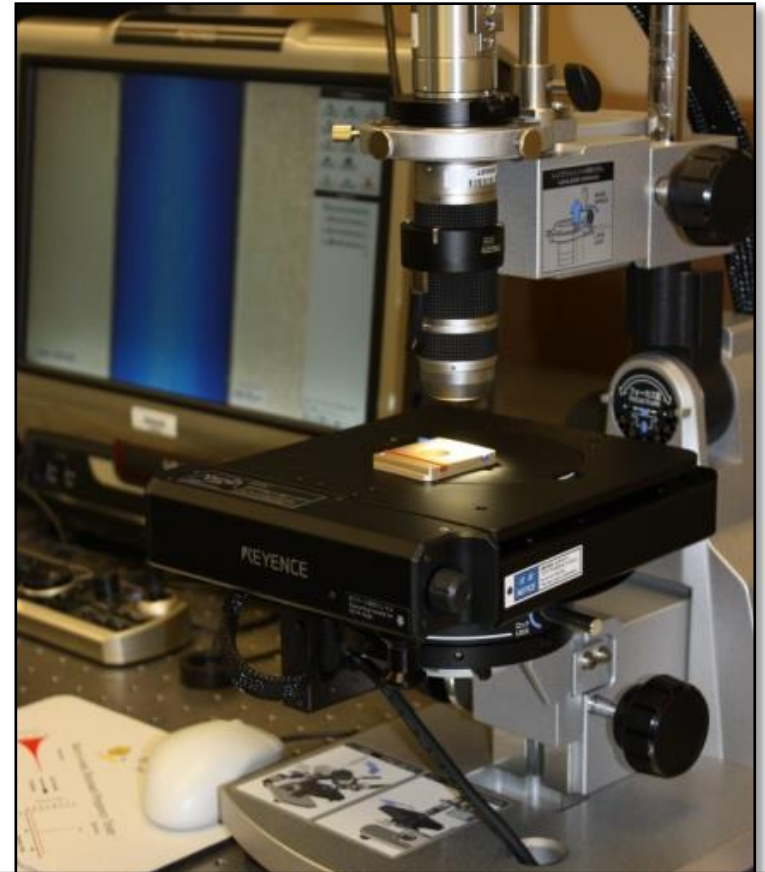




# Methodology: O-Ring and Interface Testing

## Downselection process of O-rings through rigorous testing

- **Thermal Survival Testing**
  - Compress and Test samples at  $-75^{\circ}\text{C}$
- **Low Outgassing Testing**
  - Per ASTM standards (Outsourced)
  - To protect the Optical Sensor
- **Adhesion Testing**
  - Using a Dynamic Mechanical Analyzer (DMA)
  - Constants:
    - Base-plate, compression, rate of applied force
  - Variables:
    - Interface plate, O-ring, coatings, treatments
    - Thermal Testing ( $-50^{\circ}\text{C}$ ,  $+50^{\circ}\text{C}$ )
      - For promising pairs
- **Stiffness Testing**
  - To discover force necessary to compress O-ring
- **Surface Roughness Testing**
  - Correlation between interface and adhesion



**Keyence microscope to analyze thermal survival samples**



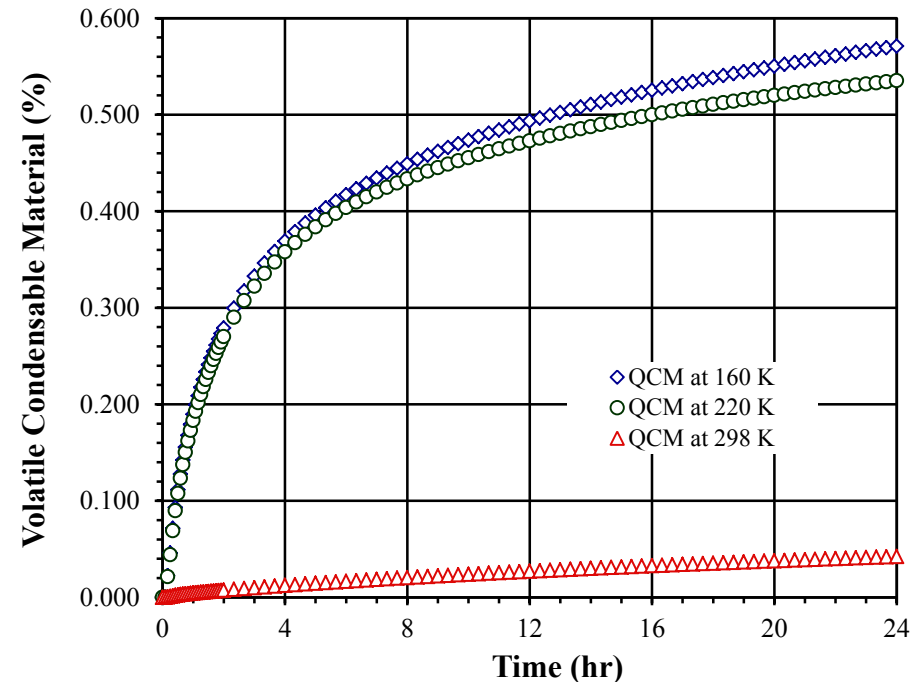


# Methodology: O-Ring and Interface Testing

## Downselection process of O-rings through rigorous testing

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    - Thermal Testing (-50°C, +50°C)
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Ja-Bar 40 O-ring Material at 125°C.

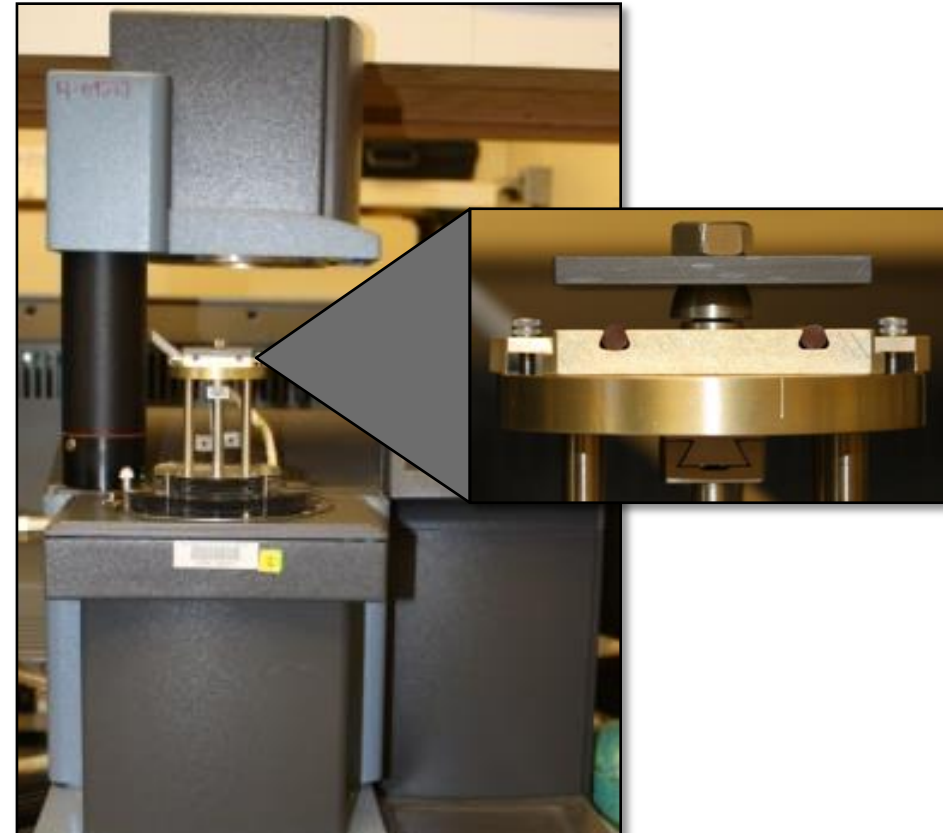




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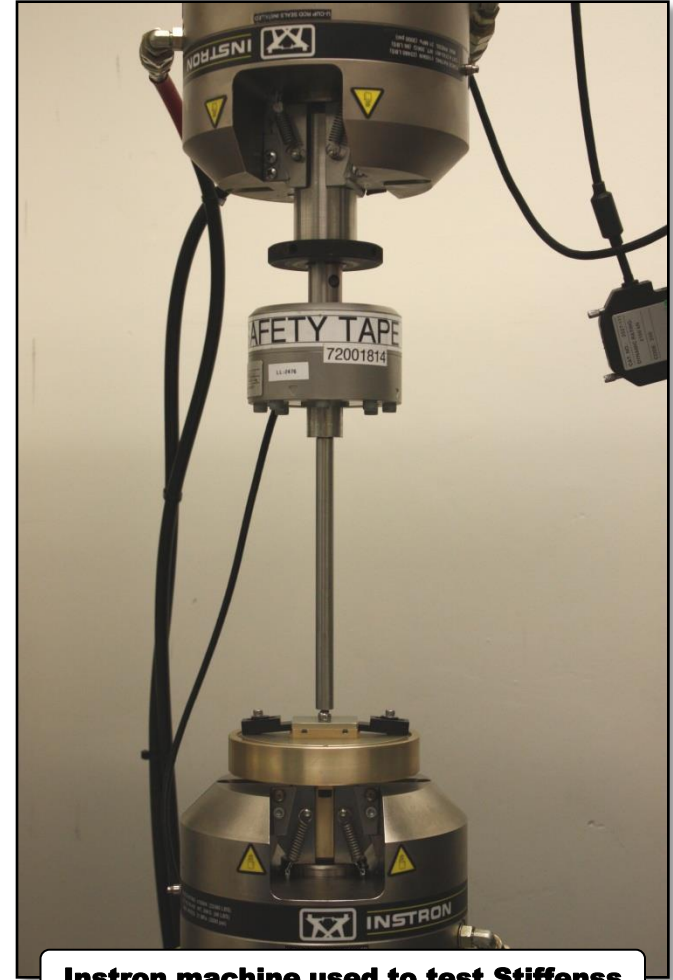
**Testing of O-Ring using the DMA**



# Methodology: O-Ring and Interface Testing

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**Instron machine used to test Stiffness**



# Methodology: O-Ring and Interface Testing

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- **Stiffness Testing**
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- **Surface Roughness Testing**
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**Zygo microscope used to measure surface roughness**



# Outline

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Introduction

**Test Results**

Part Selection

Recommendations



# Testing Outline

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Thermal Survival Testing (Pass/Fail)

Low Outgassing Testing (Pass/Fail)

Stiffness Testing (Design Related)

Adhesion Testing (Design Related)

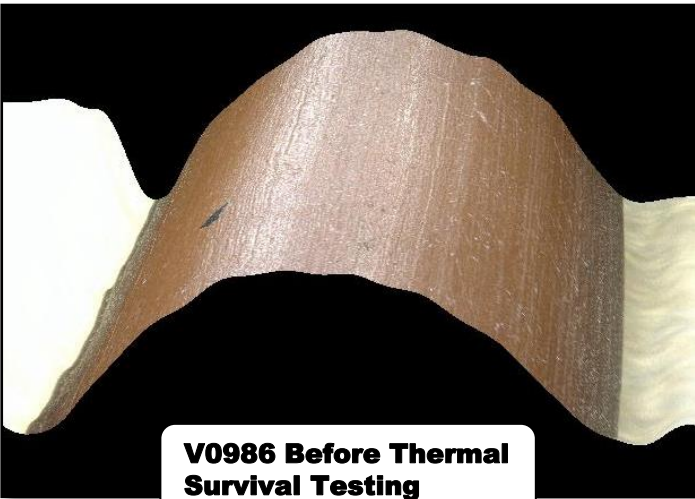
Surface Roughness Testing (Quantitative)

Thermal Adhesion Testing (Design Related)





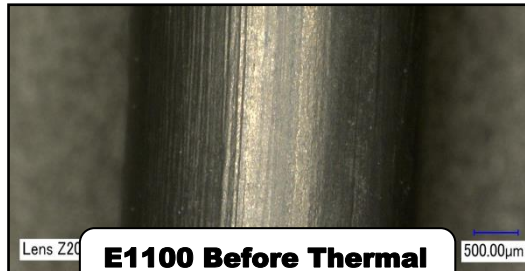
# Thermal Survival Testing



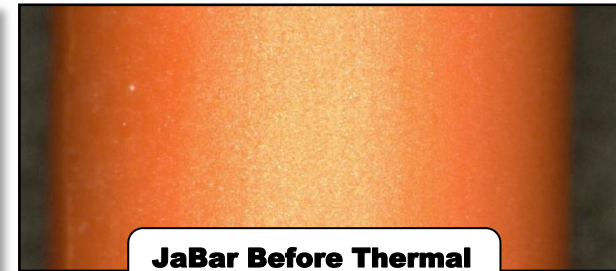
**V0986 Before Thermal Survival Testing**



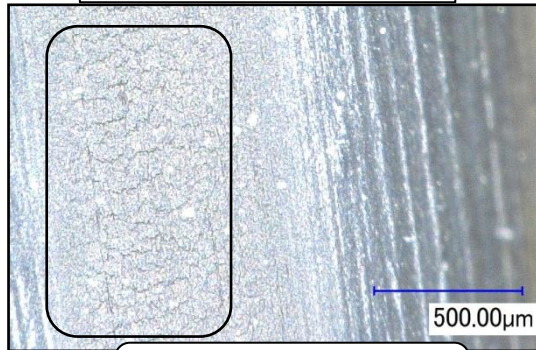
**V0986 After Thermal Survival Testing**



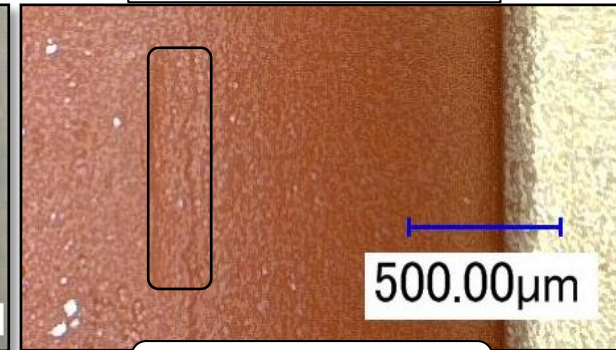
**E1100 Before Thermal Survival Testing**



**JaBar Before Thermal Survival Testing**



**E1100 After Thermal Survival Testing**



**JaBar After Thermal Survival Testing**

- Soaked at -75°C for 240 hours while under compression
- Keyence Microscope used for evaluation of cracks, pits, voids...etc.
  - Noticeable deformation in some O-rings
  - Noticeable cracks and pits were seen in E1100 and JaBar

**If the O-ring does not pass the thermal survival test, it will not be considered for space flight payloads**



# Down Selection After Thermal Survival

O-Ring	Thermal Survival	Low Outgassing	Stiffness (lbf/in <sup>2</sup> )	Adhesion (lbf/in)	Interface	Manufacturer
V0986	Pass					Parker
S0899 B2	Pass					Parker
S0899 B1	Pass					Parker
C0267	Pass					Parker
E1100	<b><u>Fail</u></b>					Parker
LM151	Pass					Parker
S0469	Pass					Parker
S0820	Pass					Parker
CV2289	Pass					MIT LL
RTV566	Pass					MIT LL
SCV2585	Pass					MIT LL
JaBar	<b><u>Fail</u></b>					Ja-Bar
S7440	Pass					Parker
S7440 FEP	Pass					MIT LL
Creavey 030	Pass					Creavey
Creavey 050	Pass					Creavey



# Testing Outline

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Thermal Survival Testing (Pass/Fail)

Low Outgassing Testing (Pass/Fail)

Stiffness Testing (Design Related)

Adhesion Testing (Design Related)

Surface Roughness Testing (Quantitative)

Thermal Adhesion Testing (Design Related)



# Low Outgassing Analysis

- **Total Mass Loss (TML)**

- < 1% Pass
- > 1% Fail

- **Collected Volatile Condensable Materials (CVCM)**

- < 0.1% Pass
- > 0.1% Fail

- C0267 and E1100 Fail
- LM151, S0469, S7440 initially fail but may pass after vacuum baking
- No results for S7440 FEP and Creavey

O-Ring	Total Mass Loss	Collected Volatile Condensable Material
V0986	0.22%	0.02%
S0899	0.10%	0.02%
C0267	8.33%	3.35%
E1100	8.82%	4.34%
LM151	1.70%	0.40%
S0469	1.66%	0.44%
S0820	0.06%	0.01%
CV2289*	0.44%	0.04%
RTV566*	0.10%	0.01%
SCV2585*	0.08%	0.01%
JaBar	0.62%	0.04%
S7440	0.31%	0.11%
S7440 FEP*		
Creavey 030		
Creavey 050		

\* Custom

**If the O-ring does not pass the outgassing test, it will not be considered for space flight payloads**



# Down Selection After Outgassing Testing

O-Ring	Thermal Survival	Low Outgassing	Stiffness (lbf/in <sup>2</sup> )	Adhesion (lbf/in)	Interface	Manufacturer
V0986	Pass	Pass				Parker
S0899 B2	Pass	Pass				Parker
S0899 B1	Pass	Pass				Parker
C0267	Pass	<b><u>Fail</u></b>				Parker
E1100	<b><u>Fail</u></b>	<b><u>Fail</u></b>				Parker
LM151	Pass	<b><u>Fail</u></b>				Parker
S0469	Pass	<b><u>Fail</u></b>				Parker
S0820	Pass	Pass				Parker
CV2289	Pass	Pass				MIT LL
RTV566	Pass	Pass				MIT LL
SCV2585	Pass	Pass				MIT LL
JaBar	<b><u>Fail</u></b>	Pass				Ja-Bar
S7440	Pass	<b><u>Fail</u></b>				Parker
S7440 FEP	Pass	n/a				MIT LL
Creavey 030	Pass	n/a				Creavey
Creavey 050	Pass	n/a				Creavey



# Testing Outline

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Thermal Survival Testing (Pass/Fail)

Low Outgassing Testing (Pass/Fail)

**Stiffness Testing (Design Related)**

Adhesion Testing (Design Related)

Surface Roughness Testing (Quantitative)

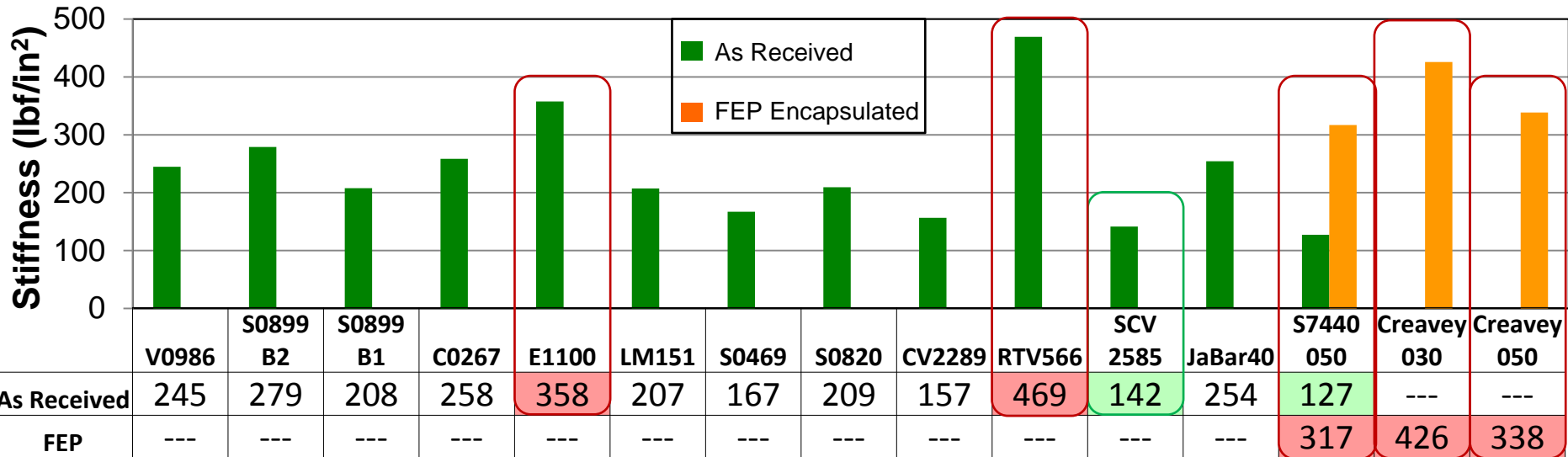
Thermal Adhesion Testing (Design Related)





# Instron Analysis for Stiffness

## O-Ring Stiffness



- Samples placed into system and the top plate was pressed down by Instron
- Force used to compress the sample and inches of deflection were recorded simultaneously
- Plots were developed for the Force (lbf) vs. the Deflection (inches)
- Linear fits were given to the plots and the slope of these plots divided by 3 inches (length of O-rings in the sample) is the Stiffness in lbf/in<sup>2</sup>

**Selecting a “soft” O-ring may be necessary in many payload applications, including ours**



# Down Selection After Stiffness Testing

O-Ring	Thermal Survival	Low Outgassing	Stiffness (lbf/in <sup>2</sup> )	Adhesion (lbf/in)	Interface	Manufacturer
V0986	Pass	Pass	245			Parker
S0899 B2	Pass	Pass	279			Parker
S0899 B1	Pass	Pass	208			Parker
C0267	Pass	<b><u>Fail</u></b>	258			Parker
E1100	<b><u>Fail</u></b>	<b><u>Fail</u></b>	<b><u>358</u></b>			Parker
LM151	Pass	<b><u>Fail</u></b>	207			Parker
S0469	Pass	<b><u>Fail</u></b>	167			Parker
S0820	Pass	Pass	209			Parker
CV2289	Pass	Pass	157			MIT LL
RTV566	Pass	Pass	<b><u>469</u></b>			MIT LL
SCV2585	Pass	Pass	142			MIT LL
JaBar	<b><u>Fail</u></b>	Pass	254			Ja-Bar
S7440	Pass	<b><u>Fail</u></b>	127			Parker
S7440 FEP	Pass	n/a	<b><u>317</u></b>			MIT LL
Creavey 030	Pass	n/a	<b><u>426</u></b>			Creavey
Creavey 050	Pass	n/a	<b><u>338</u></b>			Creavey



# Testing Outline

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Thermal Survival Testing (Pass/Fail)

Low Outgassing Testing (Pass/Fail)

Stiffness Testing (Design Related)

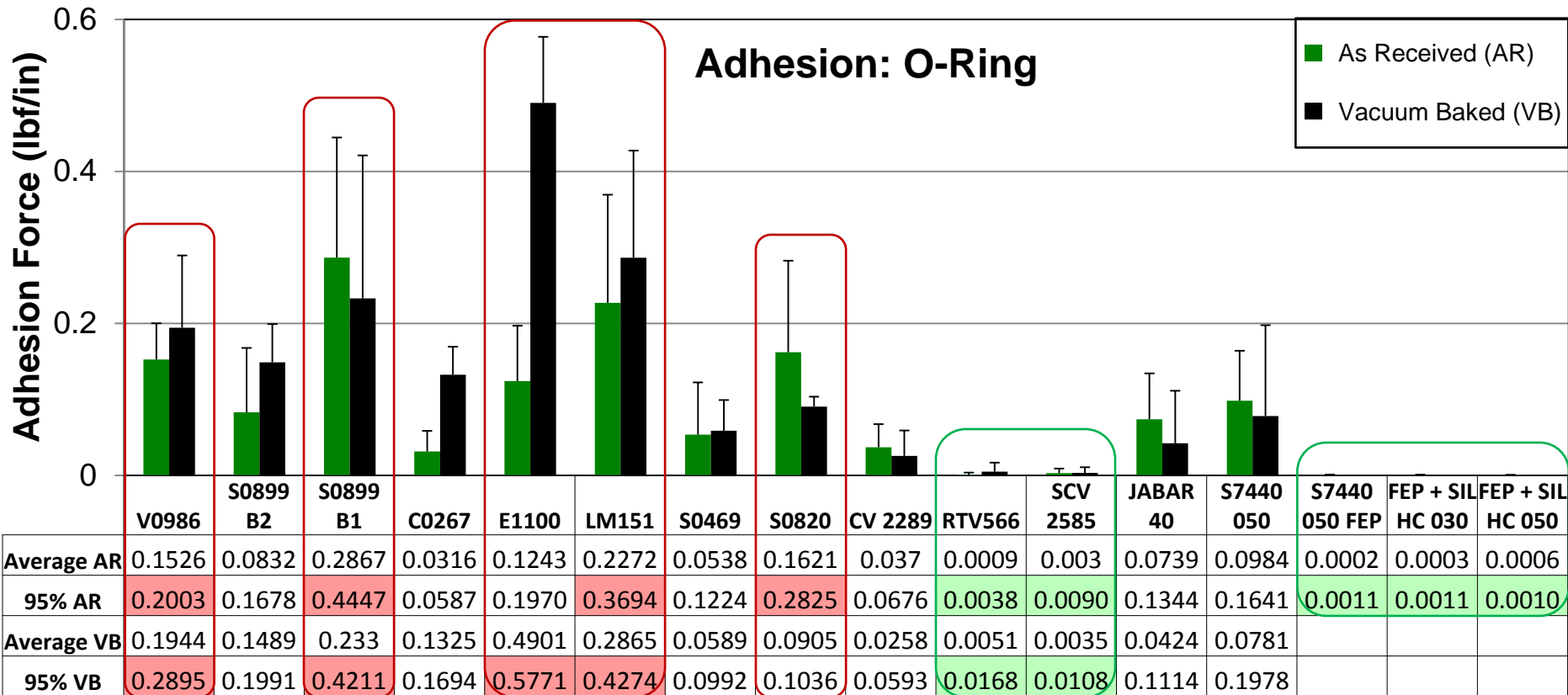
**Adhesion Testing (Design Related)**

Surface Roughness Testing (Quantitative)

Thermal Adhesion Testing (Design Related)



# Adhesion Testing



- 5-25 samples tested for each O-ring (depending on availability, previous tests, time...etc.)
- Bar represents 95% confidence level
- FEP Encapsulated O-rings resulted in practically 0 adhesion

**Some O-rings show adhesion values very close to 0 on our “typical” interface**



# Down Selection After Adhesion Testing

O-Ring	Thermal Survival	Low Outgassing	Stiffness (lbf/in <sup>2</sup> )	Adhesion (lbf/in)	Interface	Manufacturer
V0986	Pass	Pass	245	<u>0.2895</u>		Parker
S0899 B2	Pass	Pass	279	0.1991		Parker
S0899 B1	Pass	Pass	208	<u>0.4211</u>		Parker
C0267	Pass	<u>Fail</u>	258	0.1694		Parker
E1100	<u>Fail</u>	<u>Fail</u>	<u>358</u>	<u>0.5771</u>		Parker
LM151	Pass	<u>Fail</u>	207	<u>0.4274</u>		Parker
S0469	Pass	<u>Fail</u>	167	0.0992		Parker
S0820	Pass	Pass	209	0.1036		Parker
CV2289	Pass	Pass	157	0.0593		MIT LL
RTV566	Pass	Pass	<u>469</u>	0.0168		MIT LL
SCV2585	Pass	Pass	142	0.0108		MIT LL
JaBar	<u>Fail</u>	Pass	254	0.1114		Ja-Bar
S7440	Pass	<u>Fail</u>	127	0.1978		Parker
S7440 FEP	Pass	n/a	<u>317</u>	0.0011		MIT LL
Creavey 030	Pass	n/a	<u>426</u>	0.0011		Creavey
Creavey 050	Pass	n/a	<u>338</u>	0.0010		Creavey



# Testing Outline

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Thermal Survival Testing (Pass/Fail)

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Stiffness Testing (Design Related)

Adhesion Testing (Design Related)

**Surface Roughness Testing (Quantitative)**

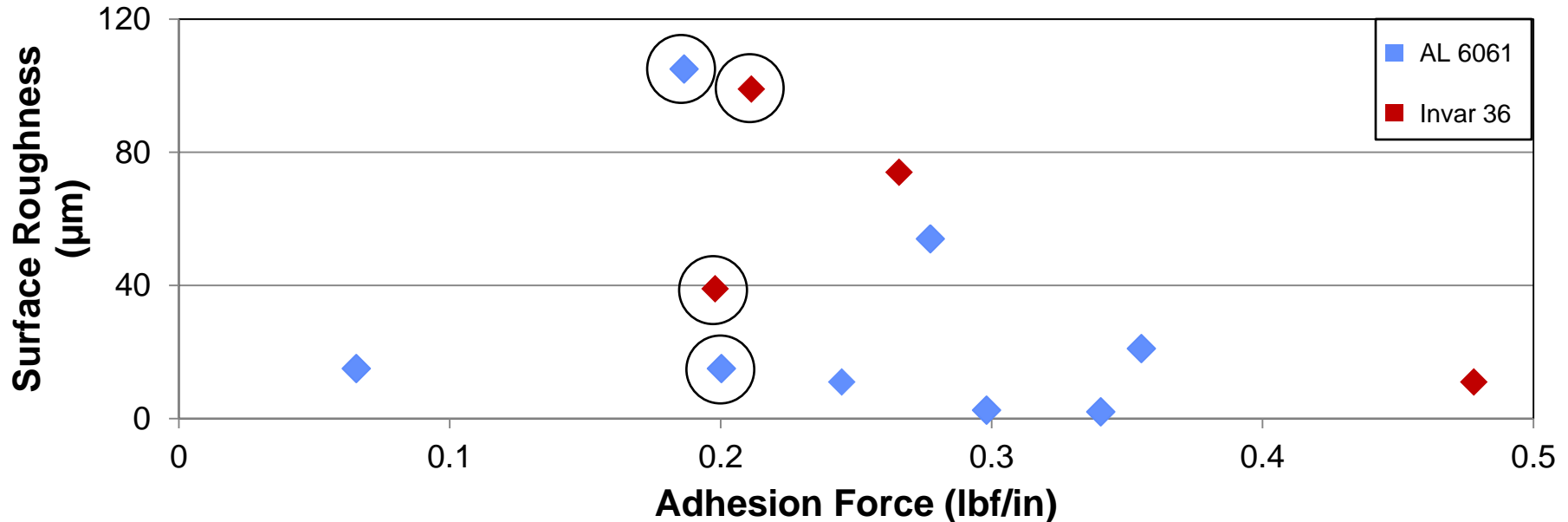
Thermal Adhesion Testing (Design Related)





# Interface's Effect on Adhesion

## Surface Roughness vs. Adhesion



- Zero adhesion not strictly possible from these interfaces
- CFM, CFB, N615, and SF2 all performed around 0.2 lbf/in
- Coatings are logistically complicated so a simple finish may be best

**There is no correlation between any surface roughness parameters and adhesion**



# Down Selection to Thermal Testing Pairs

O-Ring	Thermal Survival	Low Outgassing	Stiffness (lbf/in <sup>2</sup> )	Adhesion (lbf/in)	Interface	Manufacturer
V0986	Pass	Pass	245	<u>0.2895</u>	CFM	Parker
S0899 B2	Pass	Pass	279	0.1991	CFM	Parker
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E1100	<u>Fail</u>	<u>Fail</u>	<u>358</u>	<u>0.5771</u>	CFM	Parker
LM151	Pass	<u>Fail</u>	207	<u>0.4274</u>	CFM	Parker
S0469	Pass	<u>Fail</u>	167	0.0992	CFM	Parker
S0820	Pass	Pass	209	0.1036	CFM	Parker
CV2289	Pass	Pass	157	0.0593	CFM	MIT LL
RTV566	Pass	Pass	<u>469</u>	0.0168	CFM	MIT LL
SCV2585	Pass	Pass	142	0.0108	CFM	MIT LL
JaBar	<u>Fail</u>	Pass	254	0.1114	CFM	Ja-Bar
S7440	Pass	<u>Fail</u>	127	0.1978	CFM	Parker
S7440 FEP	Pass	n/a	<u>317</u>	0.0011	CFM	MIT LL
Creavey 030	Pass	n/a	<u>426</u>	0.0011	CFM	Creavey
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Stiffness Testing (Design Related)

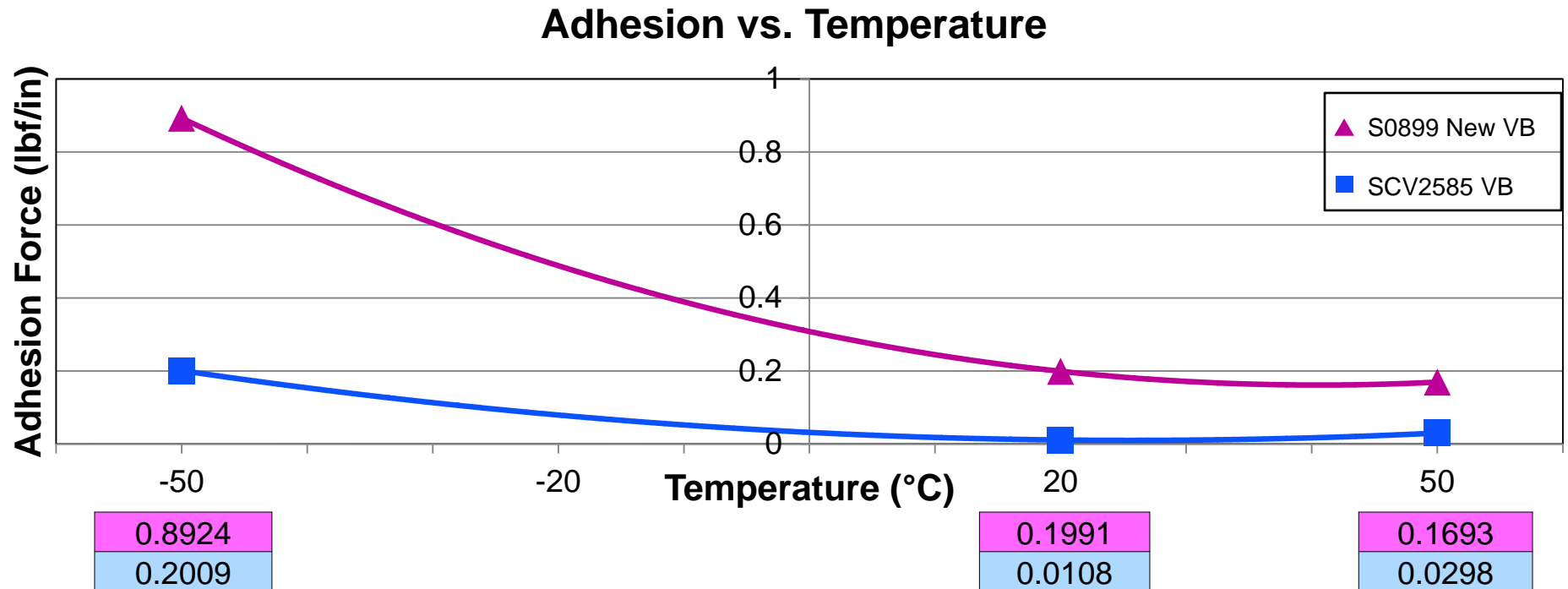
Adhesion Testing (Design Related)

Surface Roughness Testing (Quantitative)

**Thermal Adhesion Testing (Design Related)**



# Adhesion at Temperature



- Only 5 samples were tested at -50°C and +50°C
- Adhesion force increased as temperature decreased
- We project that it will increase as temperature increases as well

**Temperature has a large effect on the adhesion force of O-rings**



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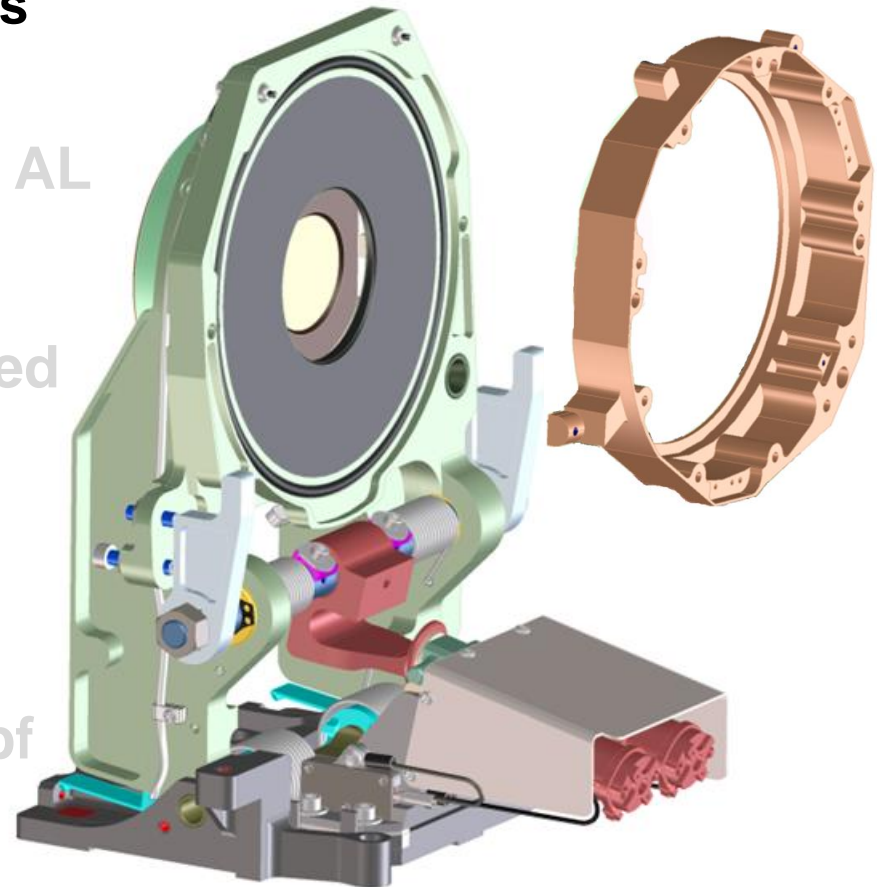
**Part Selection**

Recommendations



# Part Selection

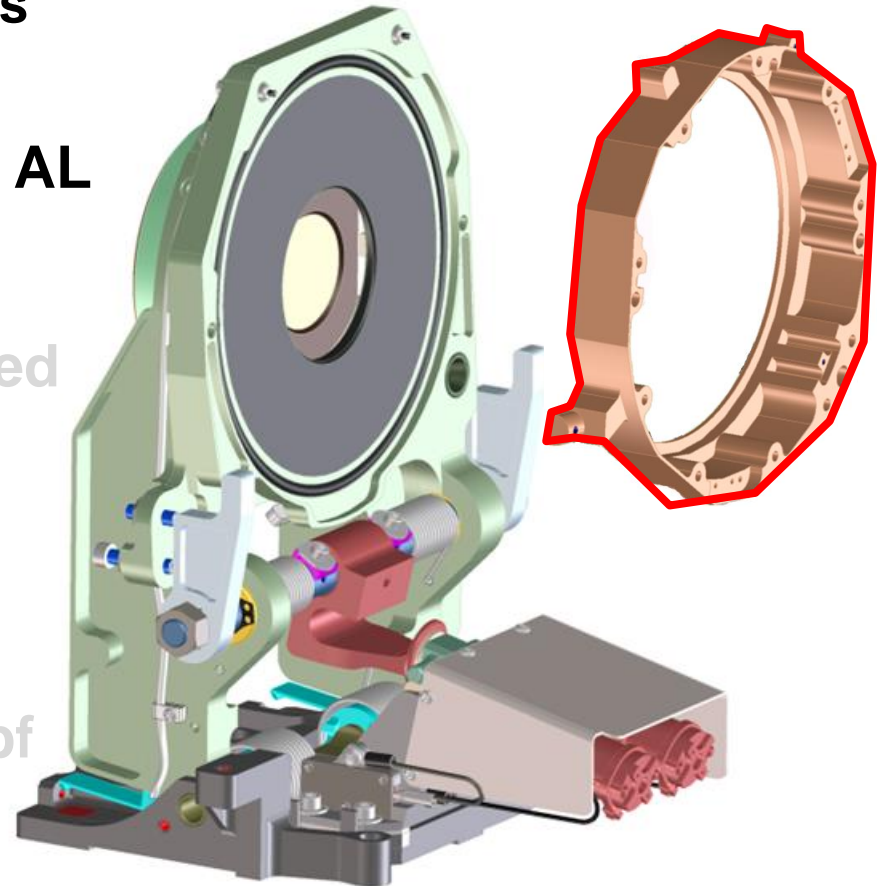
- To ensure the Launch Latch opens in orbit, 4 Launch Latch components must be specified:
- Interface – Chem Film Machined AL
- O-Ring – SCV2585 Vacuum Baked
  - Adhesion Force = 3 lbf
  - Compression Force = 27 lbf
- Upper Vlier Spring Plungers = 4.5 lbf
- Lower Torsion Springs = 5 lbf-inch





# Part Selection

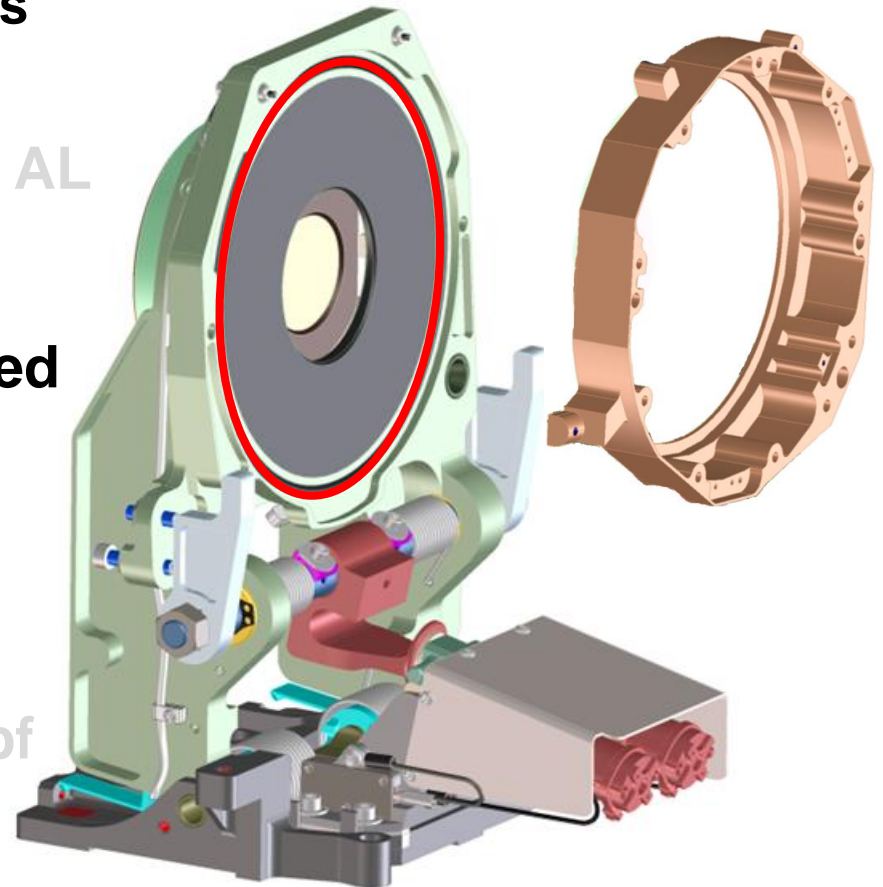
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- Lower Torsion Springs = **5 lbf-inch**

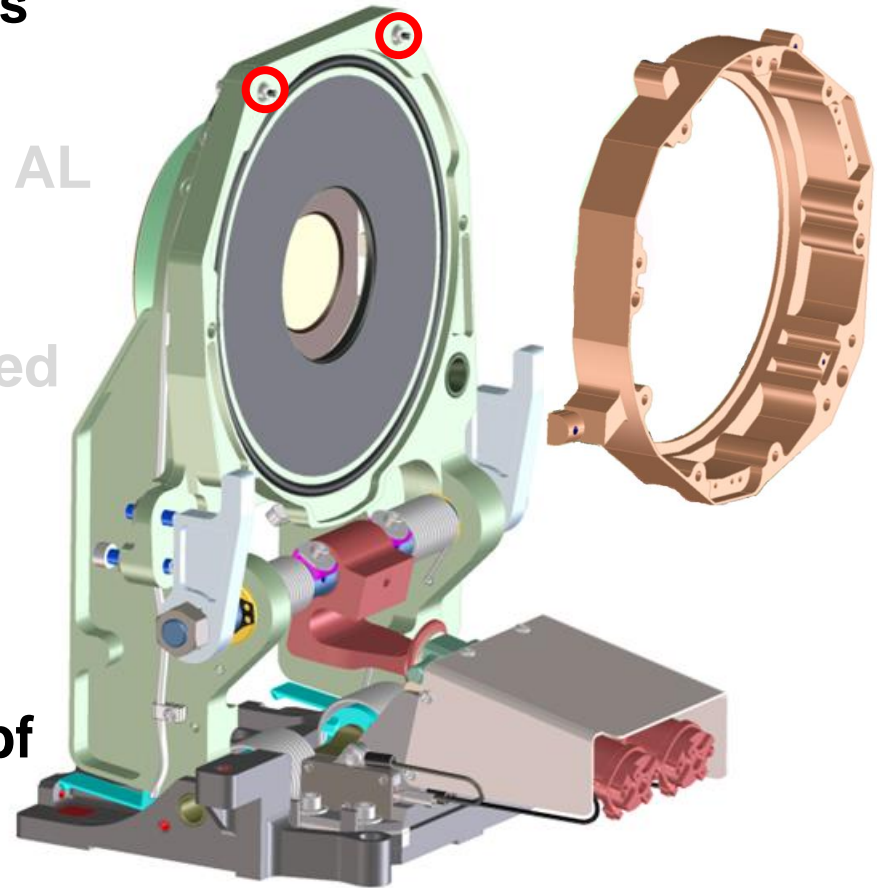






# Part Selection

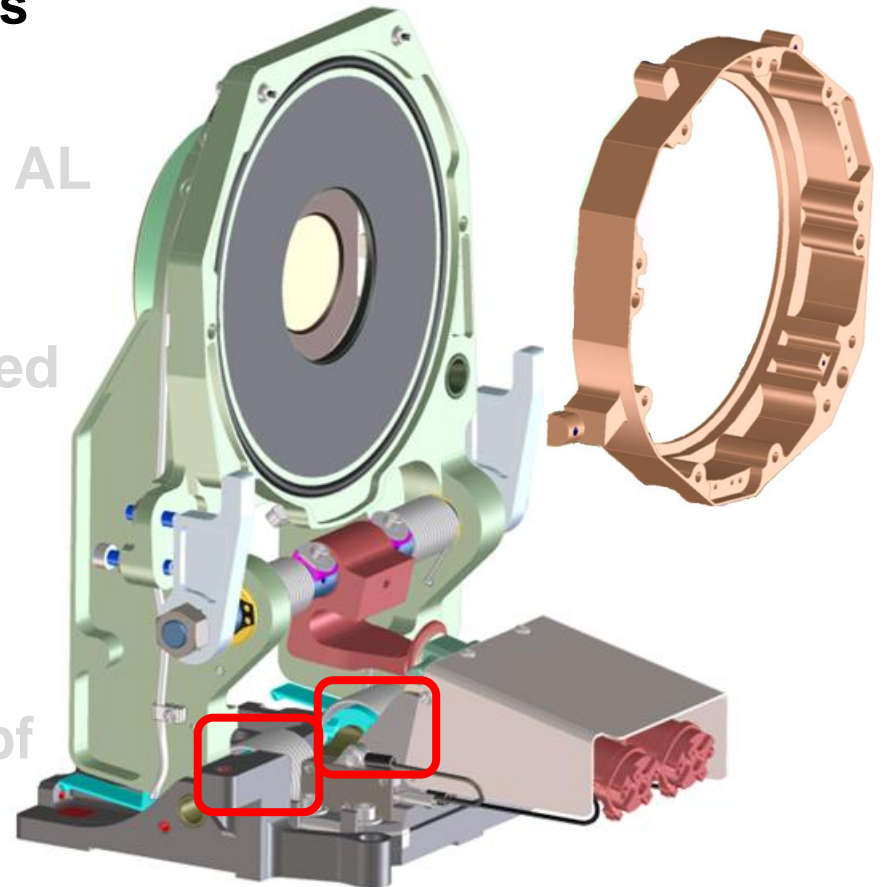
- To ensure the Launch Latch opens in orbit, 4 Launch Latch components must be specified:
- Interface – Chem Film Machined AL
- O-Ring – SCV2585 Vacuum Baked
  - Adhesion Force = 3 lbf
  - Compression Force = 27 lbf
- Upper Vlier Spring Plungers = 4.5 lbf
- Lower Torsion Springs = 5 lbf-inch





# Part Selection

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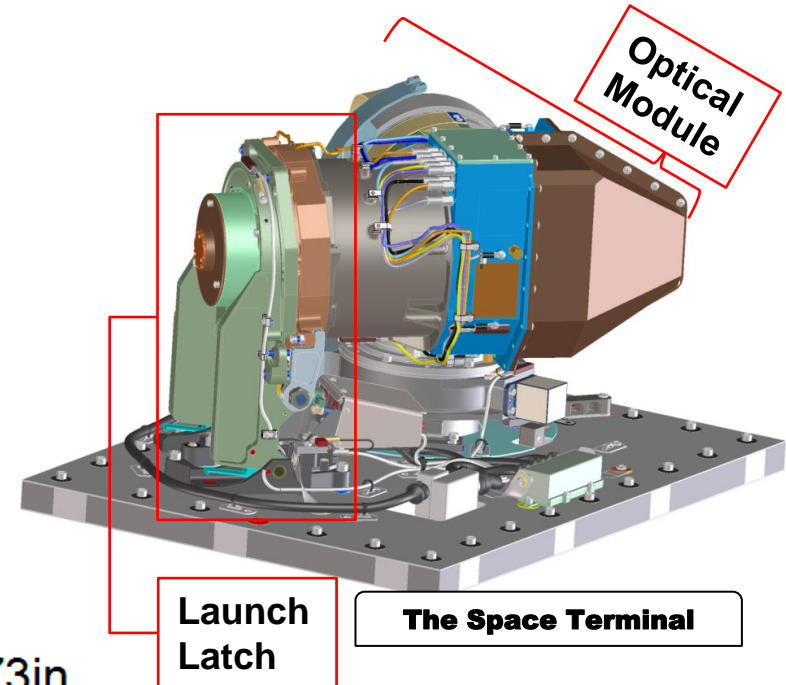
# Ensuring Latch Opens

- O-Ring adhesion force must be lower than the opening force
- After part selection all parameters are known

- $F_{\text{Adhesion}} * 1.25 < \frac{2M_{\text{LT}} + (2F_{\text{Vlier}} - P_z)r_3}{r_1}$

- $3\text{lbf} * 1.25 < \frac{2(5\text{lbf}\cdot\text{in}) + (2(4.5\text{lbf}) - 5.6\text{lbf})7.73\text{in}}{5.4\text{in}}$

- **$3.75 \text{ lbf} < 4.87 \text{ lbf}$**





# Ensuring HOPAs and Pawl Arm Survive

- **HOPA Survival**

- $F_{\text{Load-MAX}} = (260 \text{ lbf}/1.25) * 2 = 416 \text{ lbf}$
- $F_{\text{Load-MAX}} > F_{\text{Vibration}} + F_{\text{Preload}} + F_{\text{HOPA}}$
- $416 \text{ lbf} > 53 \text{ lbf} + 256 \text{ lbf} + 86 \text{ lbf}$
- **$416 \text{ lbf} > 395 \text{ lbf}$**

- **Pawl Arm Survival**

- **Finite Element Analysis**
  - Max allowable stress =  
 $150 \text{ ksi}/1.25 = 120,000 \text{ psi}$
  - Our system = **97,100 psi**



**High Output Paraffin Actuator (HOPA)**



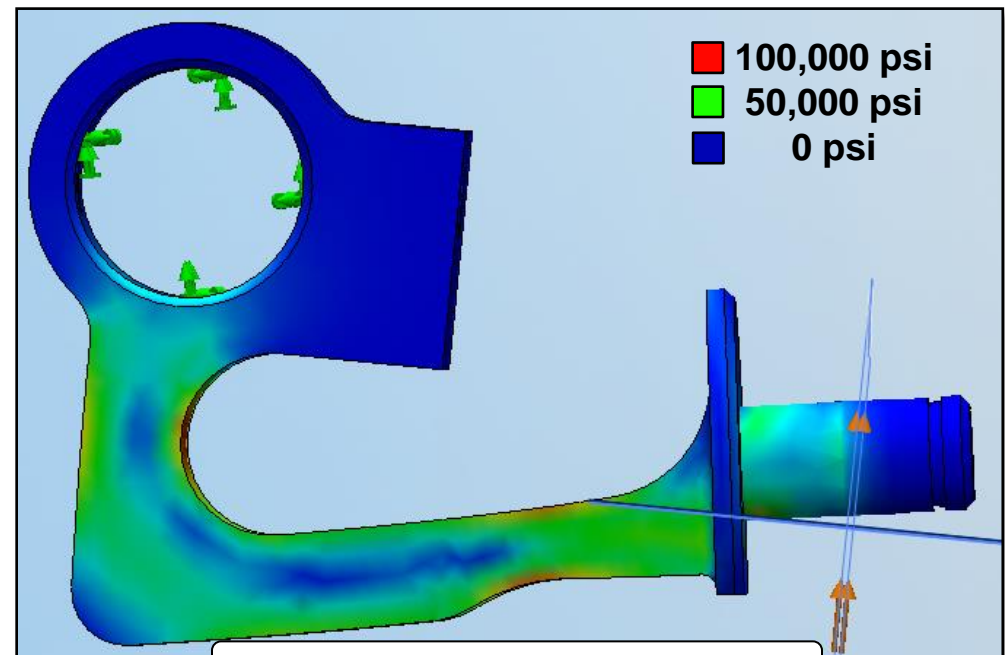
# Ensuring HOPAs and Pawl Arm Survive

- HOPA Survival

- $F_{\text{HOPA-MAX}} = 520\text{ lbf} / 1.25 = 416\text{ lbf}$
- $F_{\text{HOPA-MAX}} > F_{\text{Vibration}} + F_{\text{Preload}} + F_{\text{Close}}$
- $416\text{ lbf} > 53\text{ lbf} + 256\text{ lbf} + 86\text{ lbf}$
- $416\text{ lbf} > 395\text{ lbf}$

- Pawl Arm Survival

- Finite Element Analysis
  - Max allowable stress =  $150\text{ ksi} / 1.25 = 120,000\text{ psi}$
  - Our system = **97,100 psi**



Finite Element Analysis of Pawl Arm



# Outline

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Introduction

Test Results

Part Selection

**Recommendations**





# Recommendations for Future Work

- **Look into different treatments**
  - Interface - Thin layer of carbon
  - O-Ring - SiOx
- **Look into teflon coatings that are much softer**
  - Teflon gave us zero adhesion
  - Teflon coated samples tested so far too stiff
- **Test different O-rings on the interfaces**
  - V0986 resulted in uneven lifts
  - No clear trend with adhesion and coatings
- **Perform more thermal tests**
  - Be more confident in our adhesion results
  - Potentially find equation for adhesion in terms of temperature



**Bin of Test Samples**



**Sample During Assembly**



# Acknowledgments

We would like to acknowledge and thank the following persons for all their help in making this project a success:

From Worcester Polytechnic Institute:

- **Professor Gatsonis** – Project Advisor
- **Professor Clancy** – Site Coordinator

From MIT Lincoln Laboratory:

- **Carl Hart** – Supervisor, Group 71
- **Jesse Mills** – Supervisor, Group 75
- **Todd Mower** – Technical Staff, Group 75
- **Sharon Hardiman** – Secretary, Group 75
- **Carol Mullinax** – Secretary, Group 75
- **Peter Anderson** – Technician, Group 72
- **Russ Goodman** – Technician, Group 81
- **Scott Hillis** – Clean Room Technician, Group 72
- **Glenn Matot** – Clean Room Technician, Group 72
- **Tasha Naylor** – Clean Room Technician, Group 72

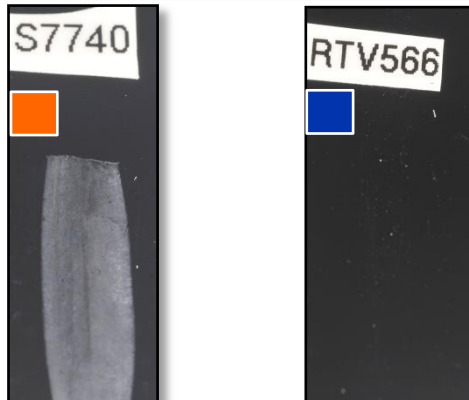




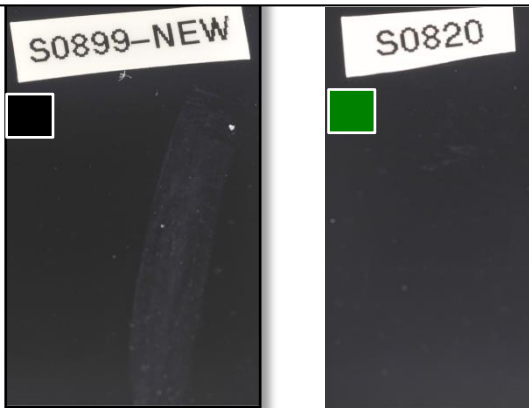
# Questions?



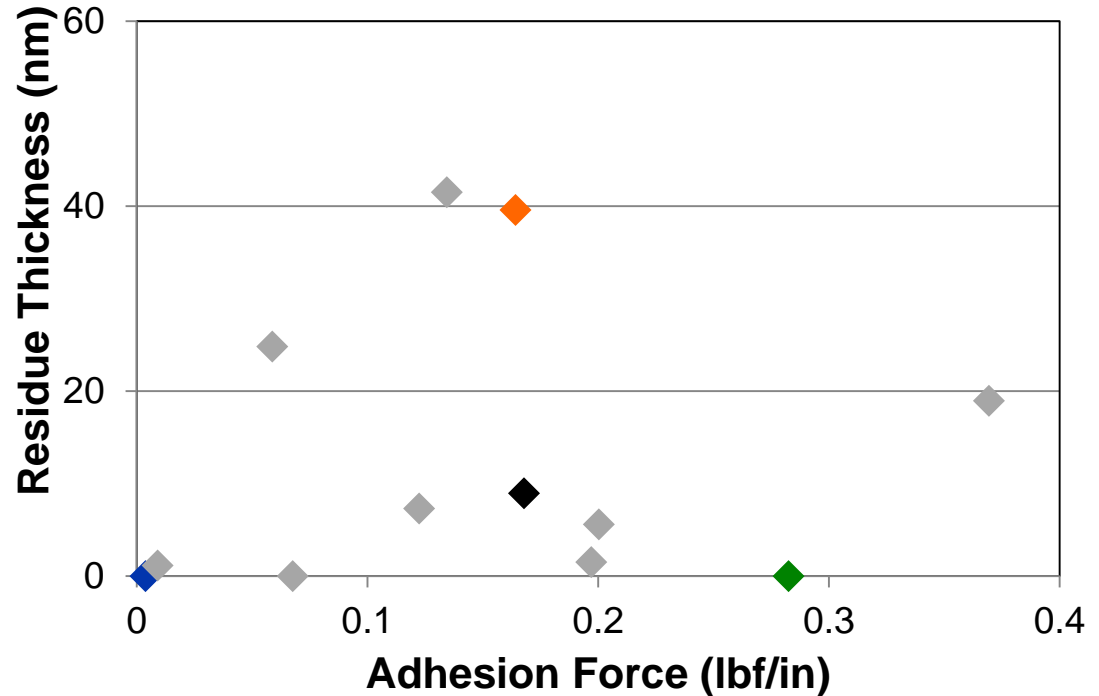
# Residue's Effect on Adhesion



Squish Test Residue of O-Ring Samples



Residue vs. Adhesion

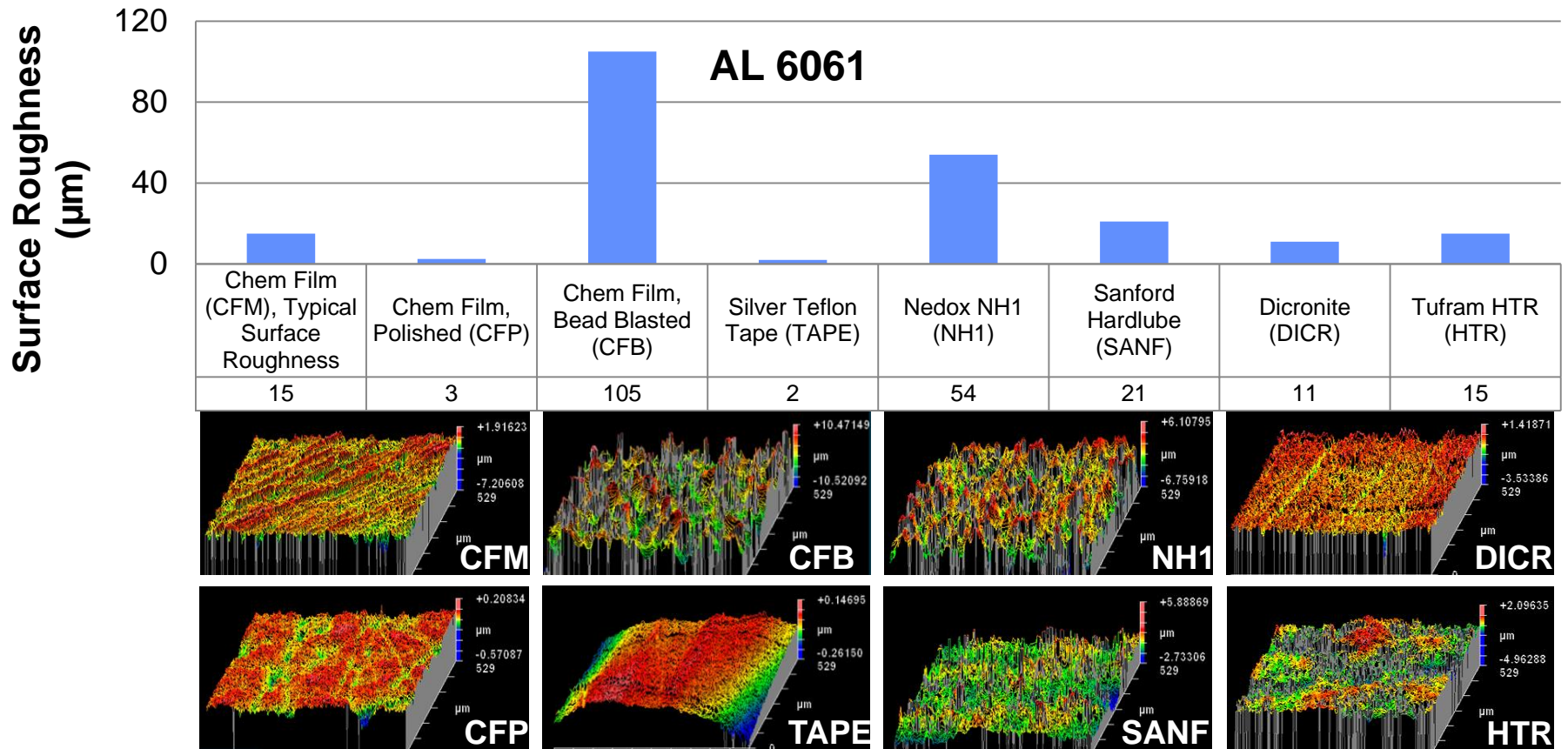


- S7740 and S0899-New (B2) have a similar adhesion force, but a large range in residue thickness
- RTV566 and S0820 have residue values of zero, but a large range in adhesion force

**There does not appear to be any correlation between residue thickness and adhesion**



# Surface Roughness: AL 6061

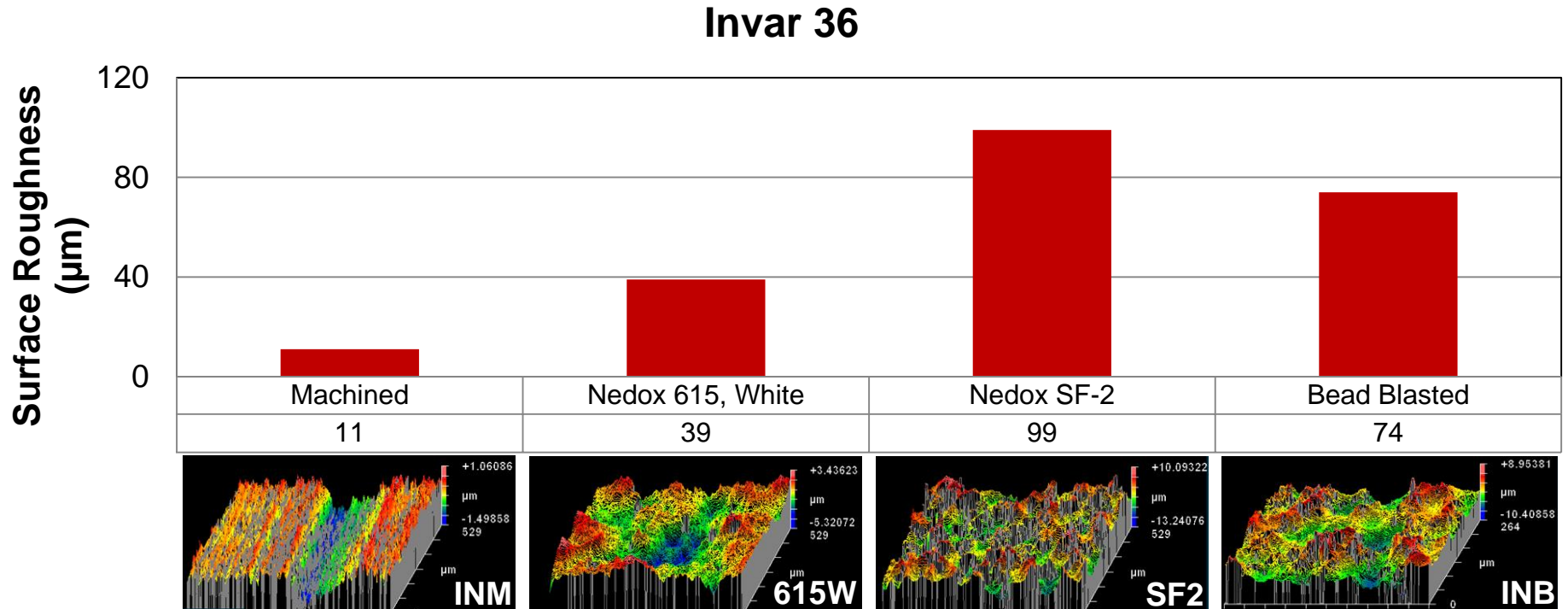


- Used Zygo to test for average surface roughness among other surface parameters
- Range from 2  $\mu\text{m}$  (TAPE) to 105  $\mu\text{m}$  (CFB)

The interface surface roughness of our interfaces may have an effect on the adhesion of the O-ring



# Surface Roughness: Invar 36

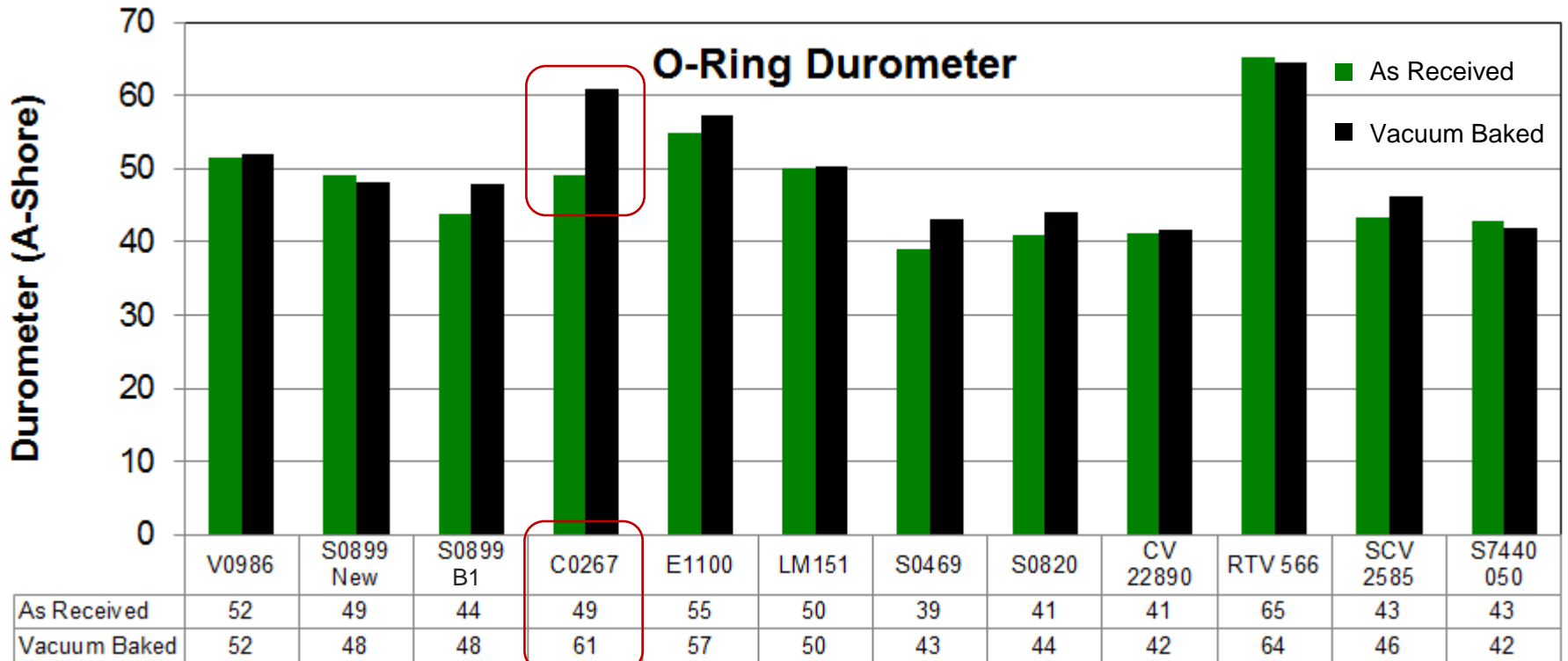


- Range from 11  $\mu\text{m}$  (Machined) to 99  $\mu\text{m}$  (SF2)
- Parameters of RMS, and Max Height show a similar order in increasing values for both Invar 36 and AL 6061

**Other surface roughness parameters of our interfaces may have an effect on the O-ring adhesion**



# Durometer Testing

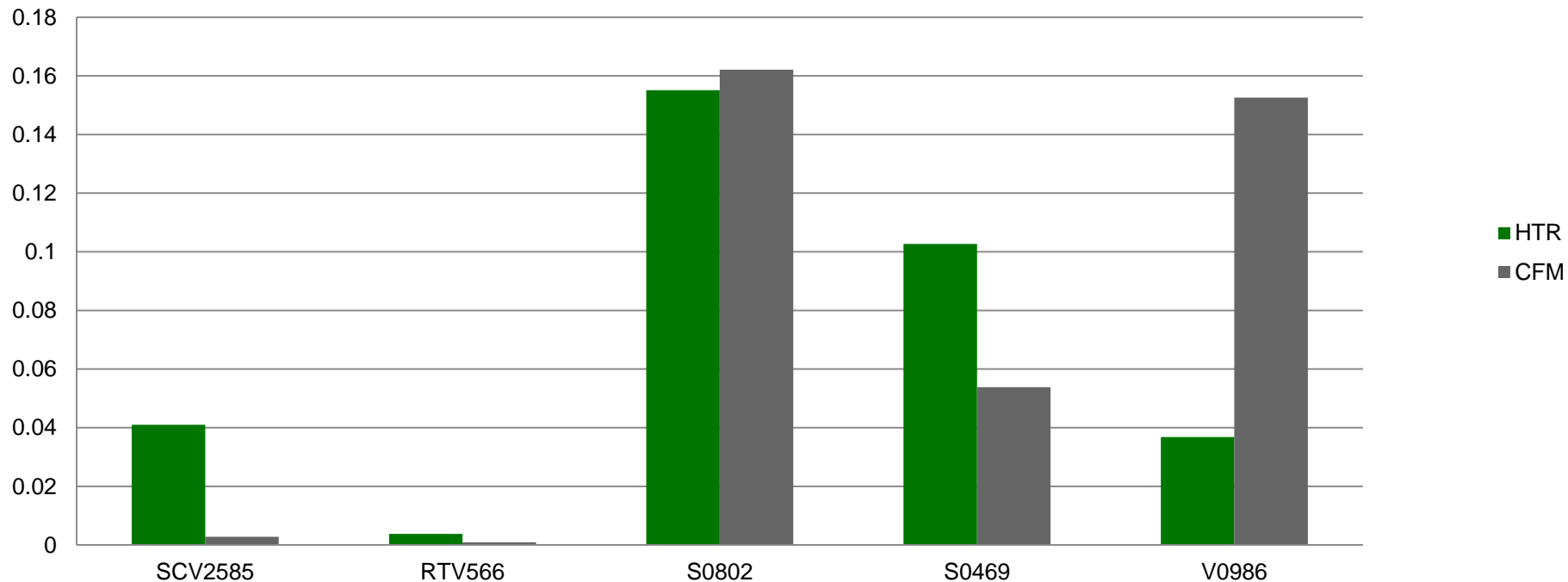


- Durometer taken before and after vacuum baking
- Most O-ring has slight increase in durometer
- C0267 increased by 12 while no others increased by more than 4
  - Equivalent of changing from the hardness of a rubber stamp to wiper blades



# Adhesion on HTR and CFM

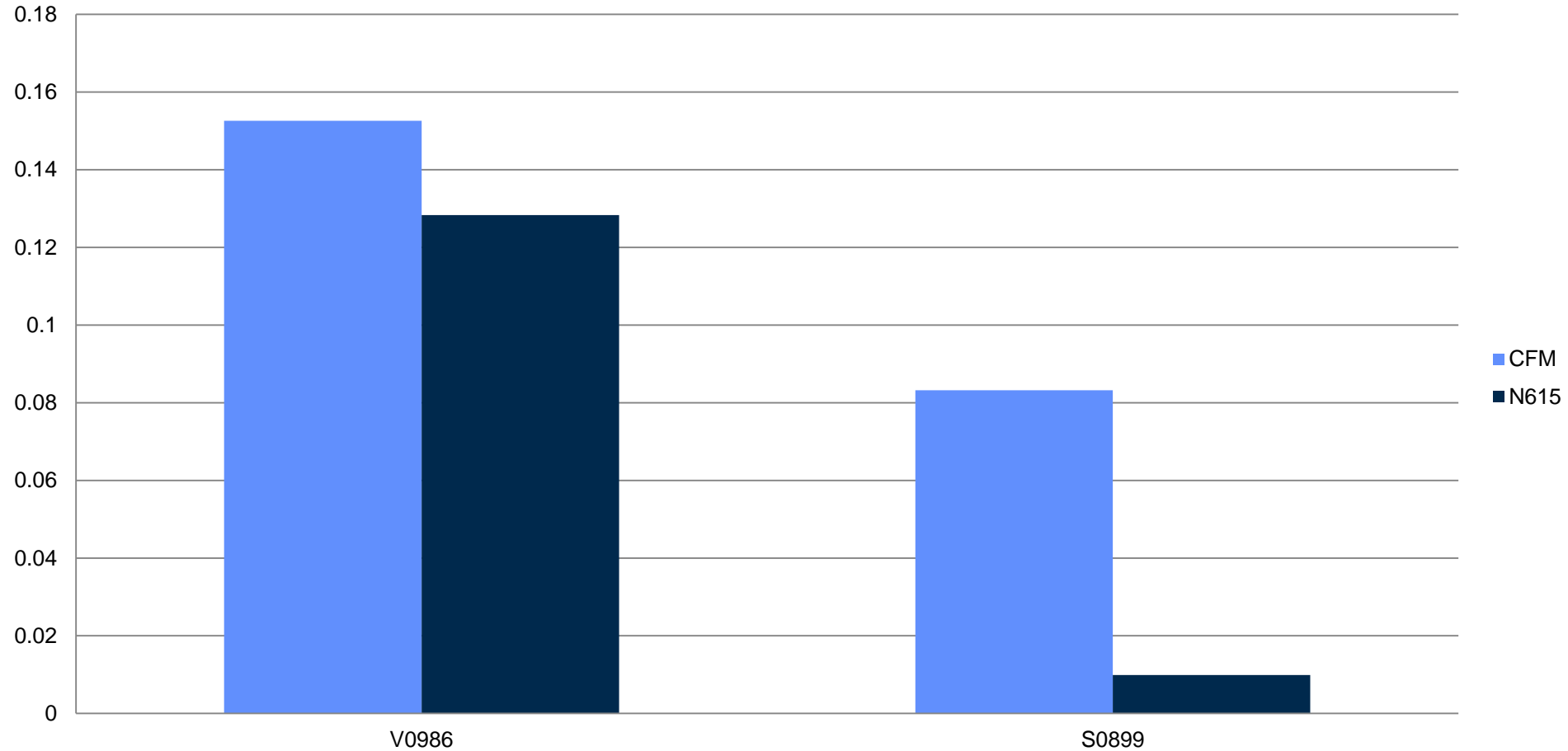
## Adhesion (lbs/inch) vs. O-ring and Surface Type





# Adhesion of CFM and N615

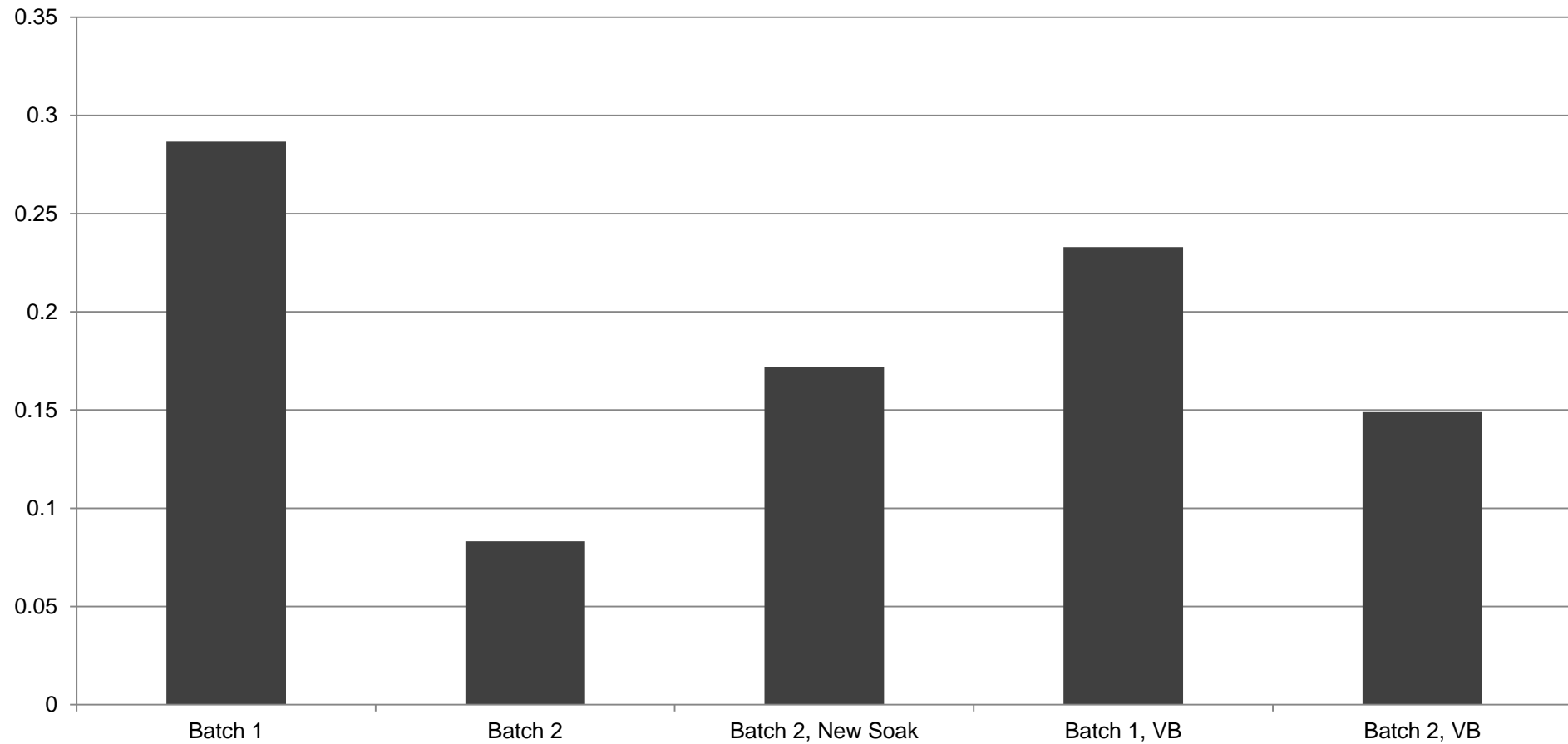
## Adhesion (lbs/inch) vs. O-ring and Surface Type





# S0899 Adhesion

## S0899 Adhesion for Different O-ring Batches







# Stiffness vs Durometer

